



Microprocessor Controlled Aerial Robotics Team (MicroCART)

Dr. Jones and Dr. Elia

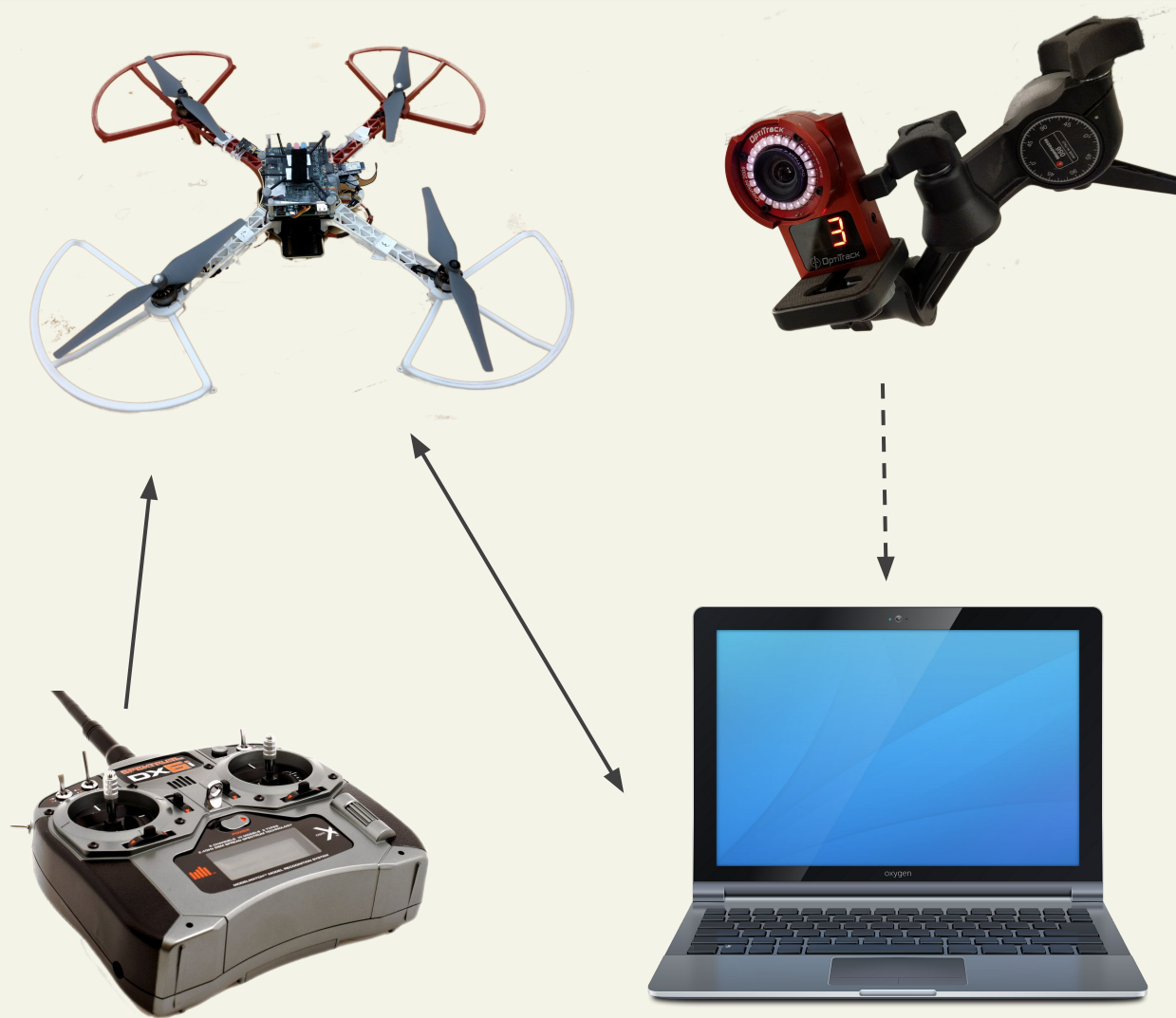
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Purpose of MicroCART

To develop an aerial robot as a research platform for controls and embedded systems.

System Overview



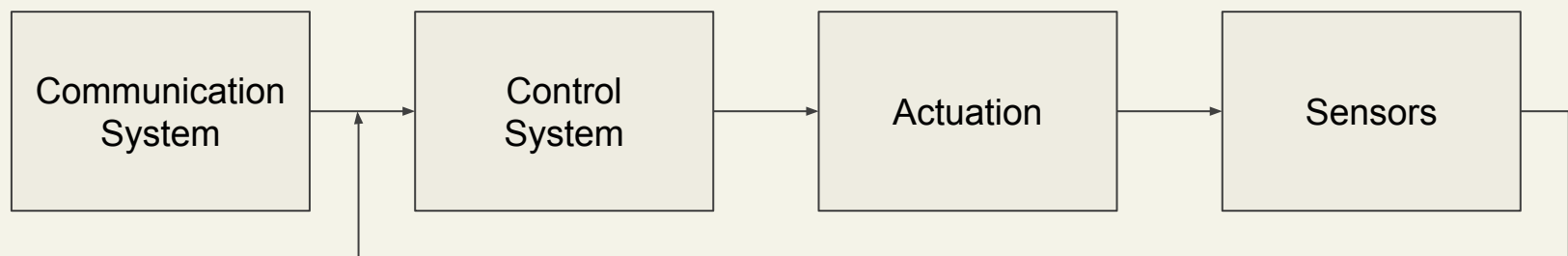
Goals and Deliverables

- Improved Flight Ability
 - Autonomous Flight
 - Controller designed from mathematical model
 - User-specified waypoints
 - Outdoor flight
- Modular Research Platform Features
 - Customizable controls structure
 - Flexibility in client types (GUI or CLI)
- Increased Robustness of System
 - Continuous Integration and Dedicated Hardware Tests
 - Communication reliability and throughput

Increased Flight Ability: Mathematical Model

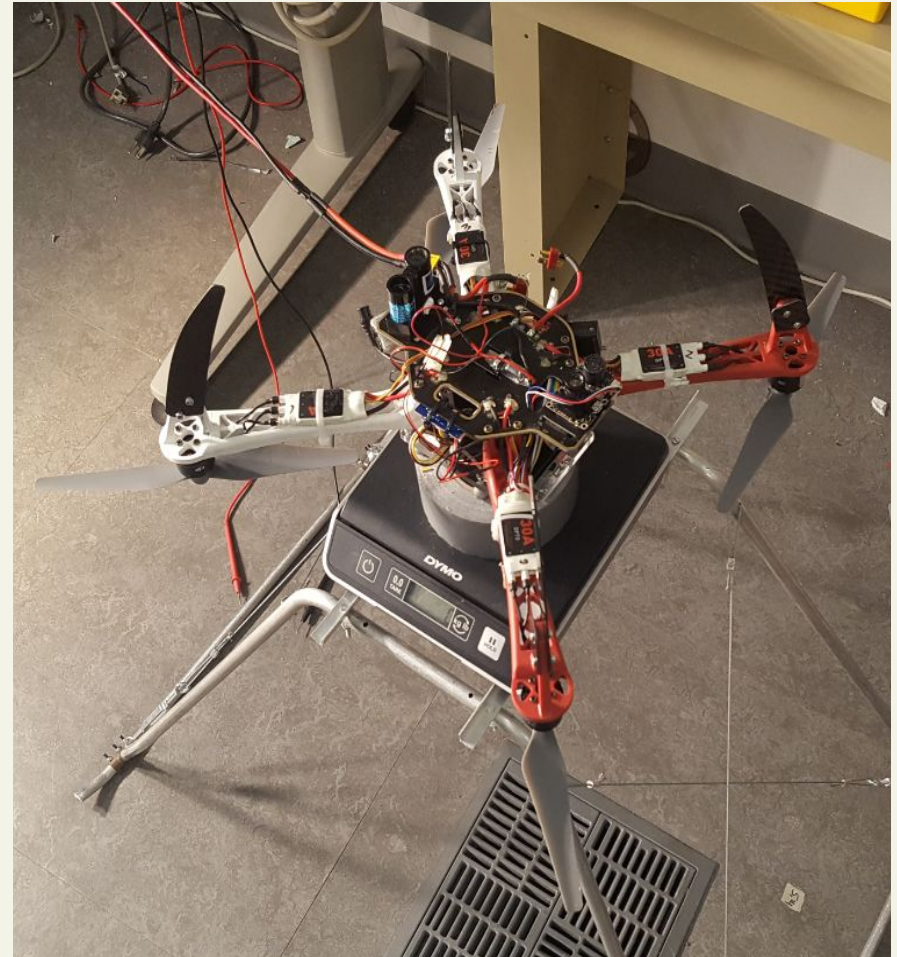
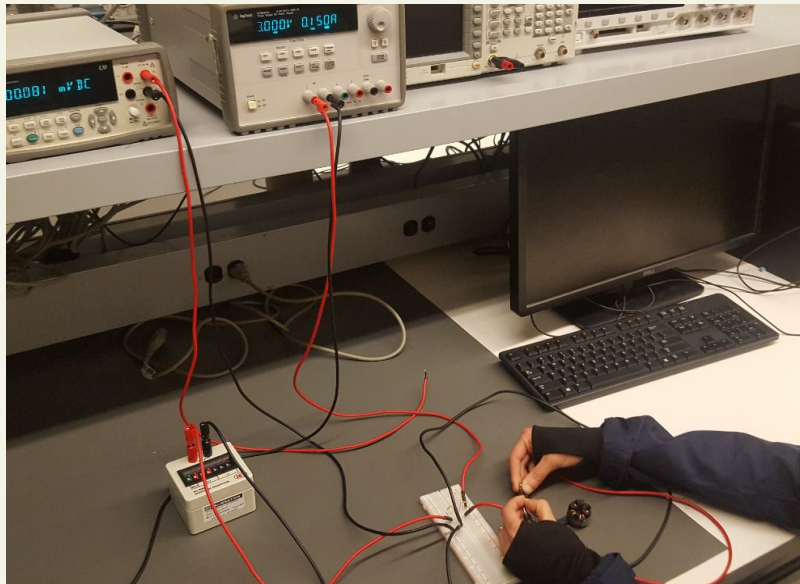
- Previously had no model of current system
 - Advantages:
 - Faster control structure development
 - Allows teams to find stabilizing controllers quickly
 - Different control structures can be simulated before being applied
 - Possibility for more advanced control in the future
 - Model based controllers can be explored

Quadcopter Model High Level Structure



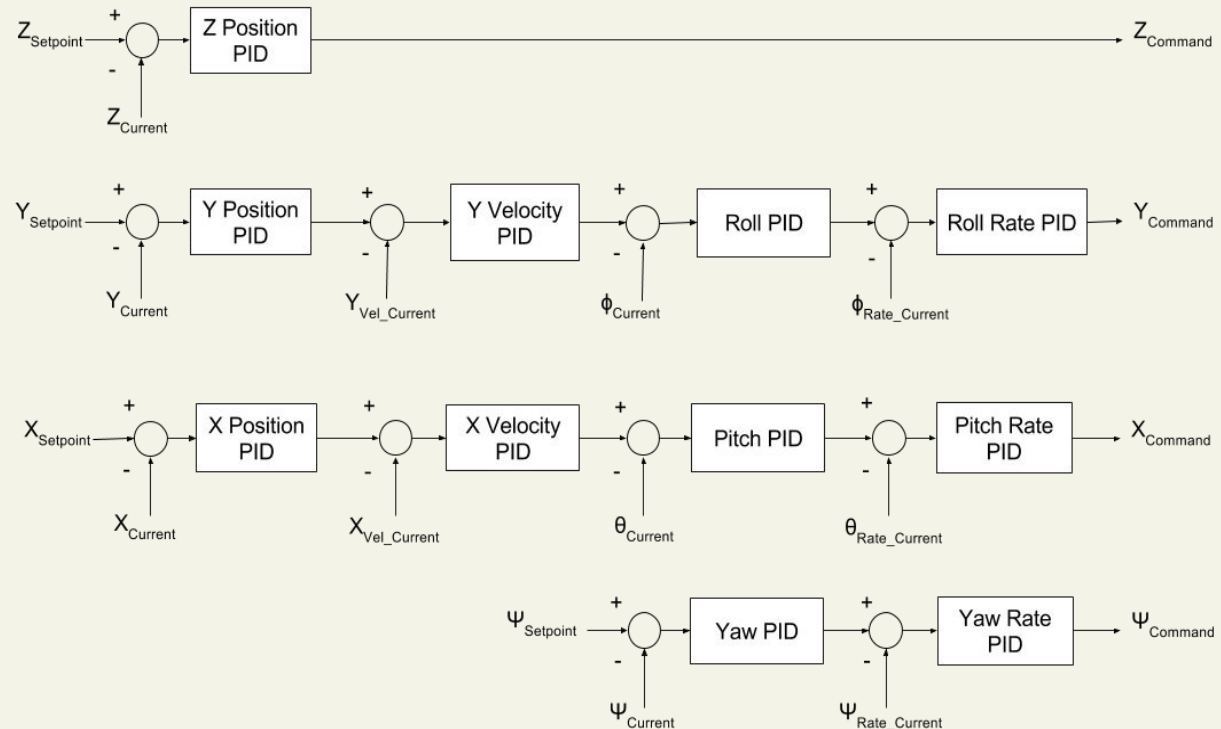
Increased Flight Ability: Creating the Model

- System Identification
- Parameters Measured:
 - Moments of inertia
 - Thrust and drag constants
 - Sensor noise characteristics
 - Motor resistance



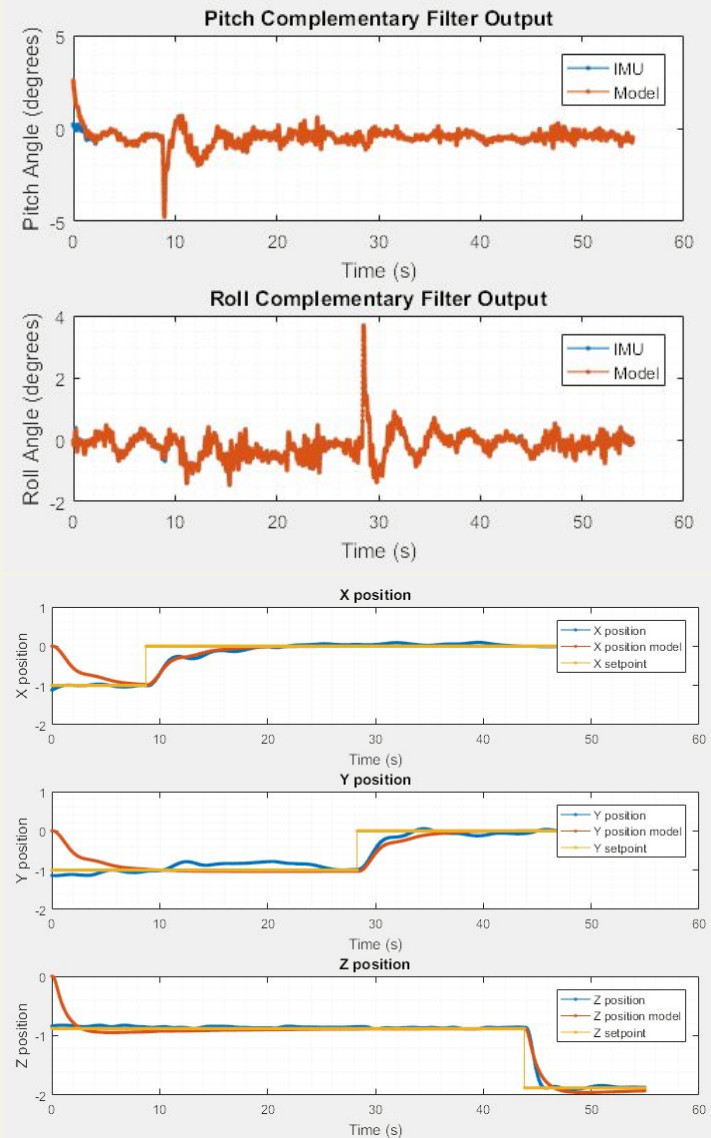
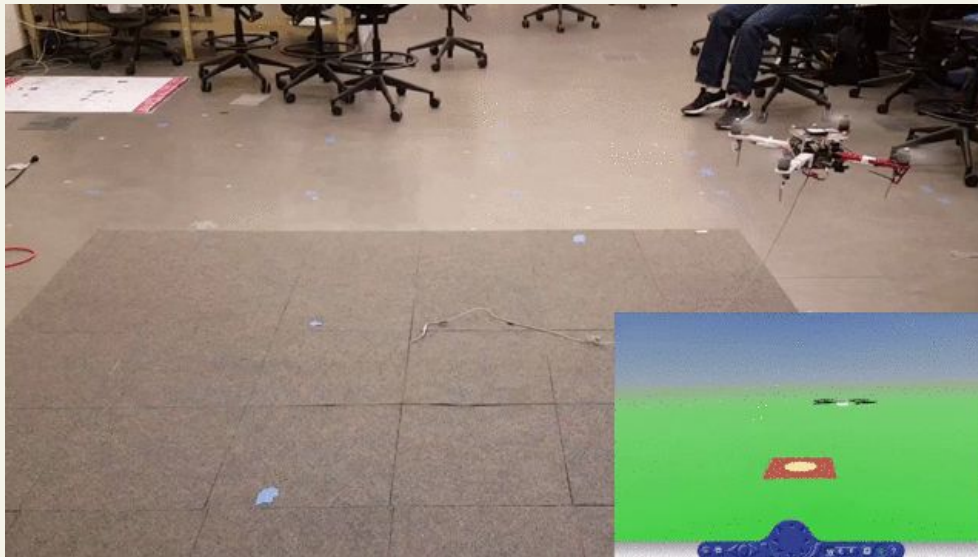
Increased Flight Ability: Control Structure

- 4 movement options
 - Height
 - Longitudinal
 - Lateral
 - Yaw
- Nested PID Structure
- Position and Velocity Control
- Euler angle and rate control



Current Model Developments

- Logging Analysis
- Setpoint Testing
 - Current model accurately reflects movement from real quadcopter

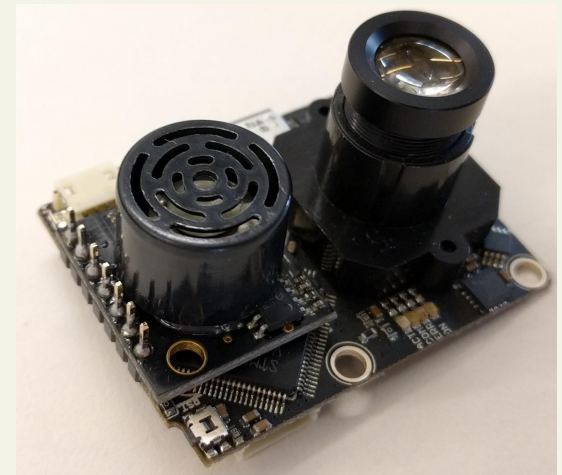


Increased Flight Abilities: Flying Outside

- Flying Outside:
 - LiDAR sensor for distance from ground
 - 1cm resolution
 - Sensor fusion algorithm combines LiDAR and accelerometer data
 - Optical Flow sensor
 - Takes high-speed images of the ground and computes pixel flow
 - Quad computes ground velocities and integrates to estimate position



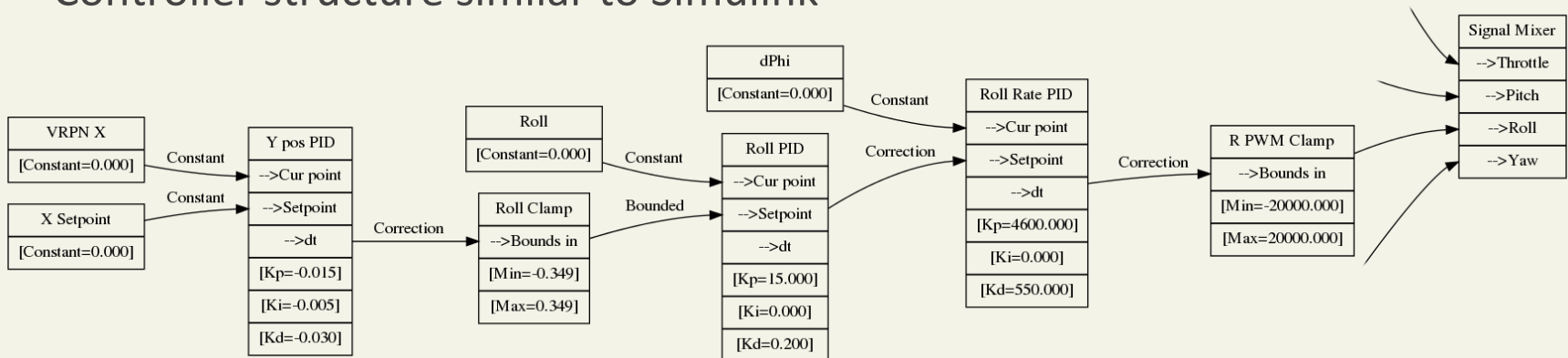
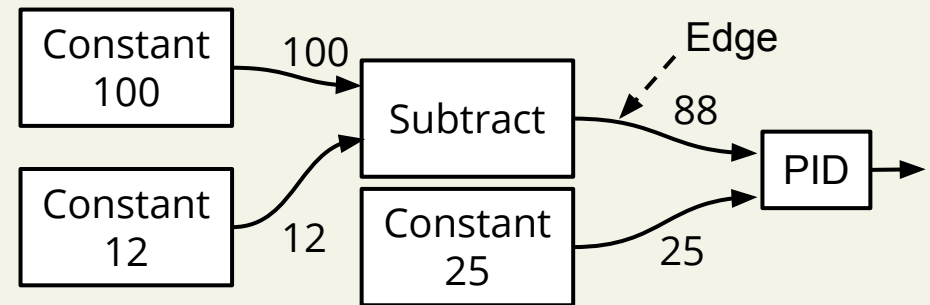
LiDAR Sensor



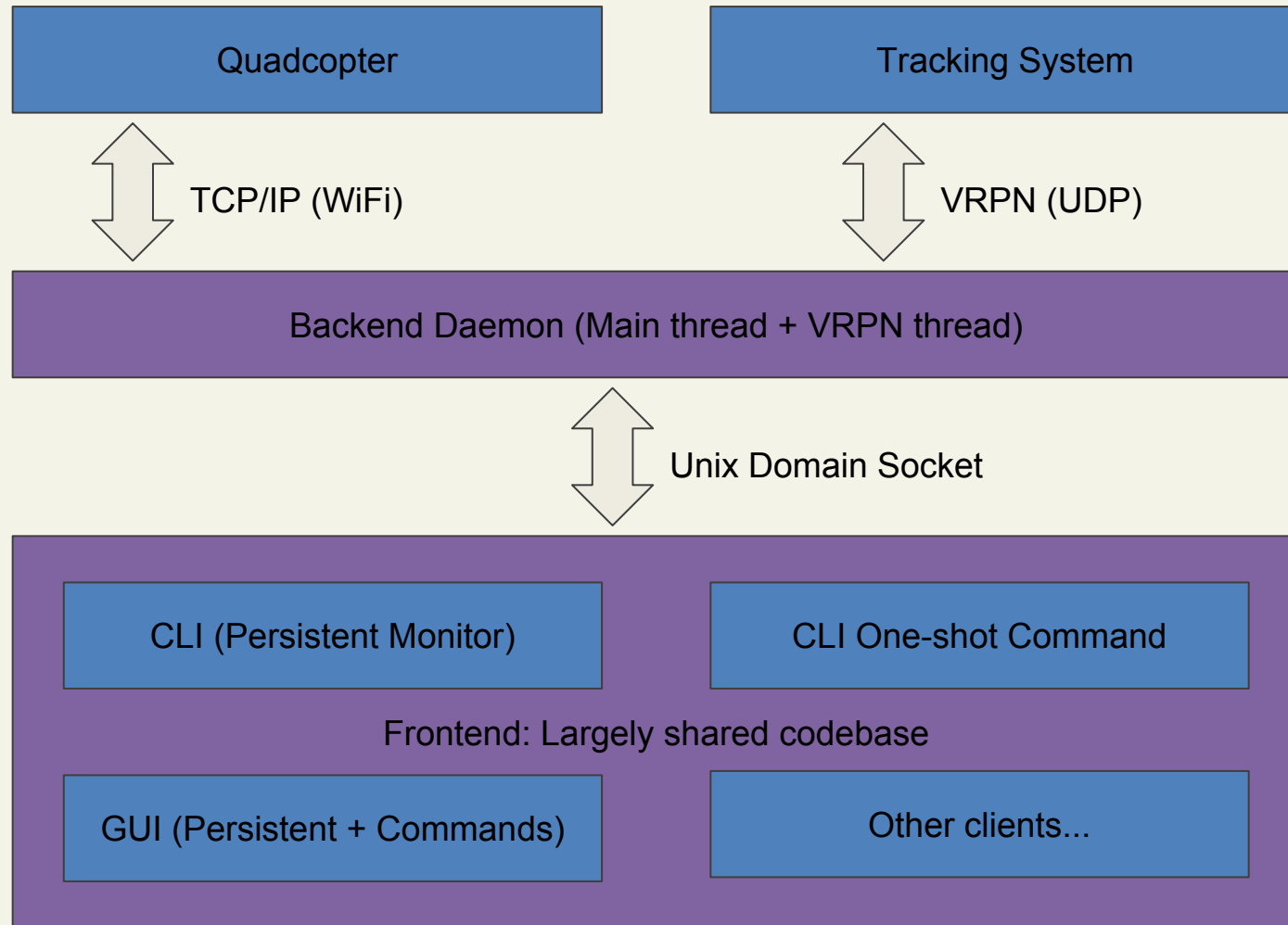
Optical Flow Sensor

Modular System: Customizable Control Structure

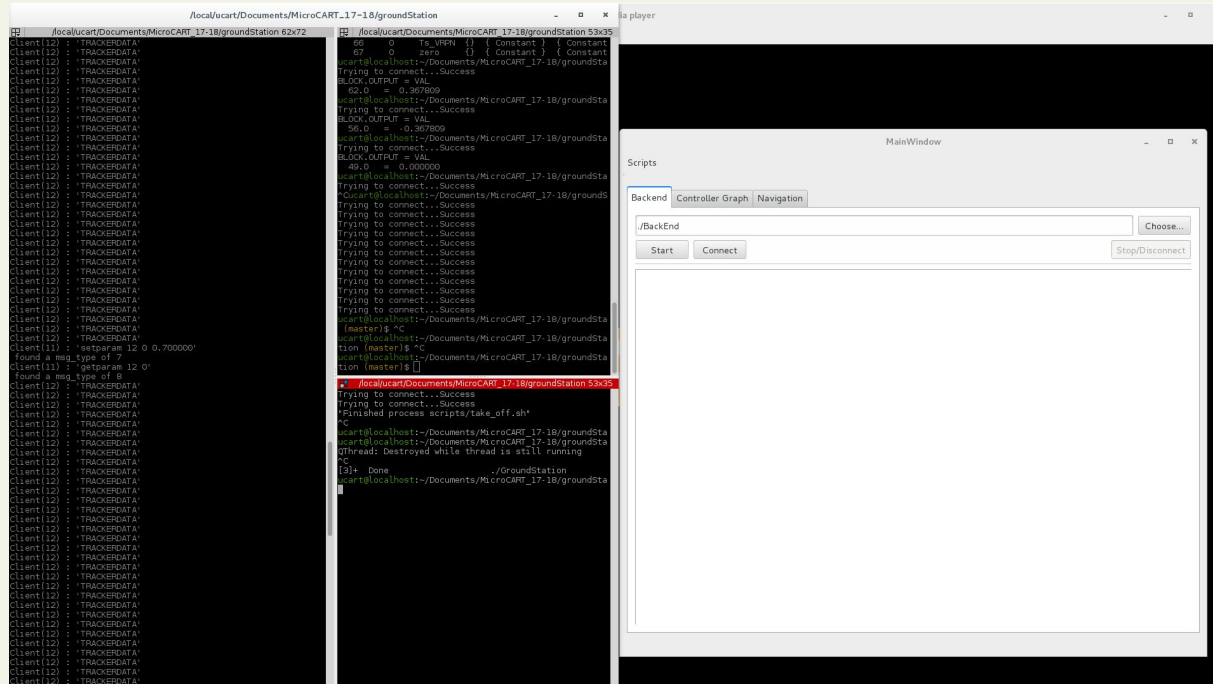
- Structure controller as a directed graph
 - Nodes are discrete functions
 - Calculated values are passed along edges to inputs of other nodes
- Benefits
 - Blocks can be developed and tested independently of the quadcopter system
 - Allows changing controller at runtime
 - Controller structure similar to Simulink



Modular System: Ground Station Architecture



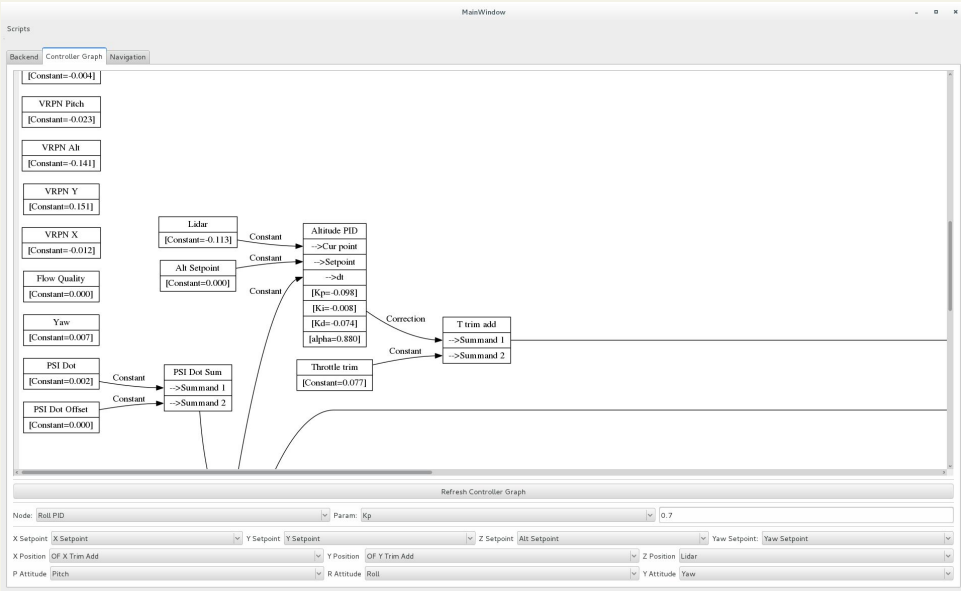
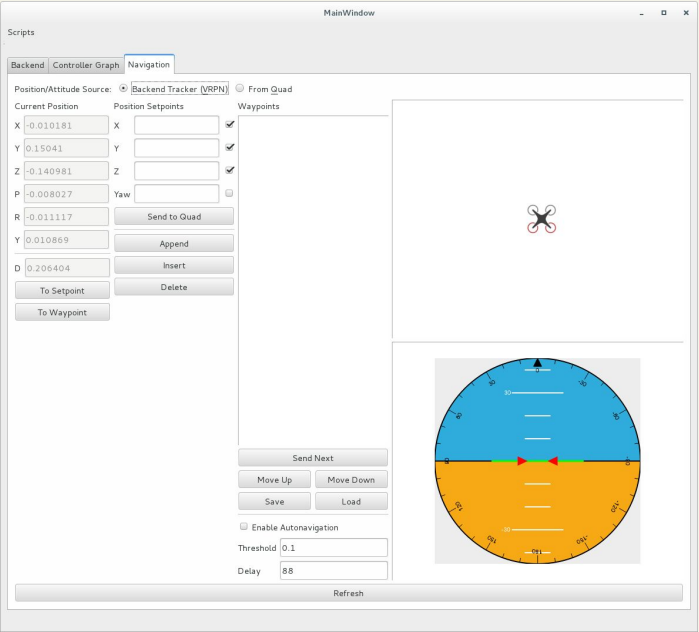
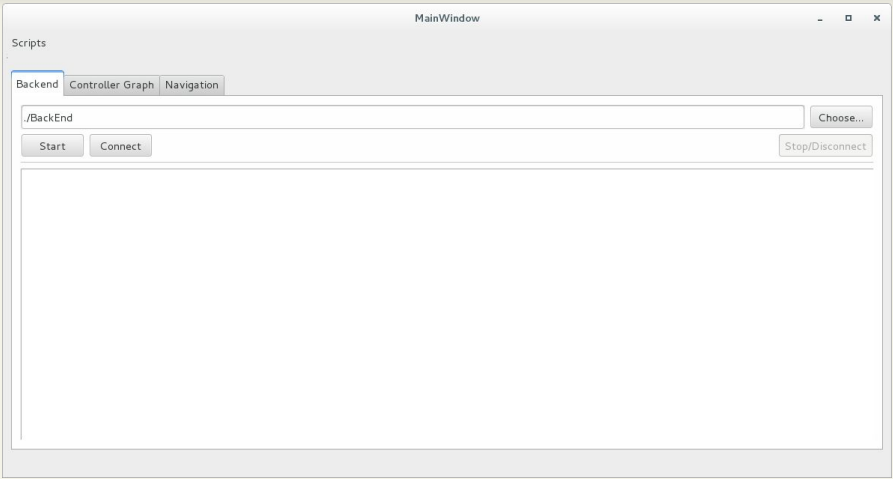
Ground Station Modular Structure



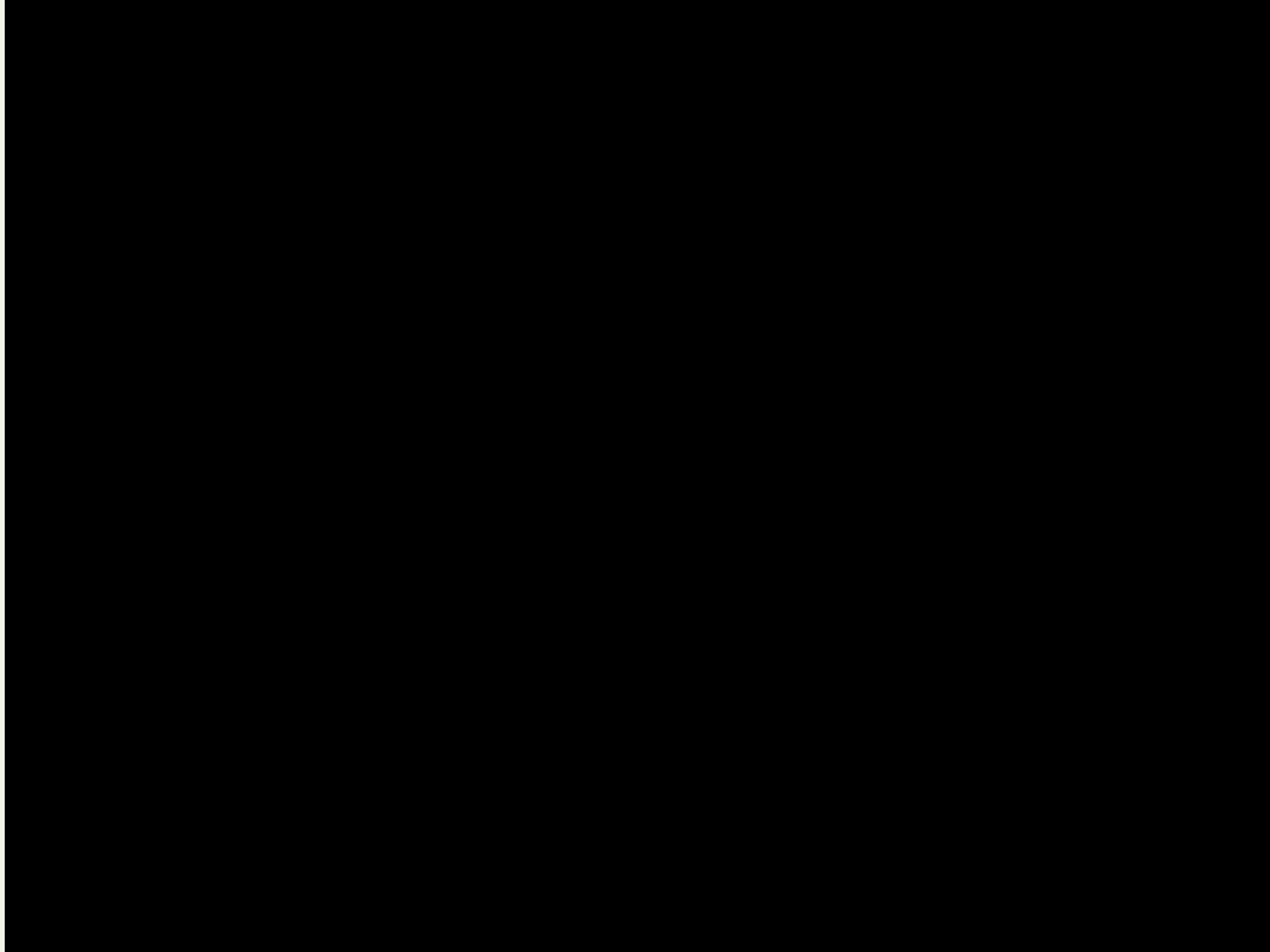
The screenshot displays the Ground Station Modular Structure. On the left, a terminal window shows a series of network logs between a client and a server, including 'Trying to connect...Success' and 'setparam 12 0 0.700000'. On the right, a GUI window titled 'MainWindow' is visible, featuring a 'Scripts' tab and buttons for 'Start', 'Connect', and 'Stop/Disconnect'.

- Backend Daemon
 - Manages quad connection, tracking system
 - Services requests from frontend
- Decoupled Command Line Interface (CLI)
 - getoutput, getparam, getsource
 - setparam, setsource, getnodes
- Intuitive Graphical User Interface (GUI)
 - Same features as CLI
 - More information at-a-glance

Ground Station GUI



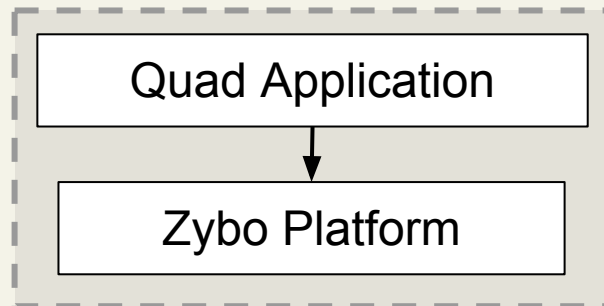
Ground Station GUI



Robustness: Improved Testing Strategy

- Problem

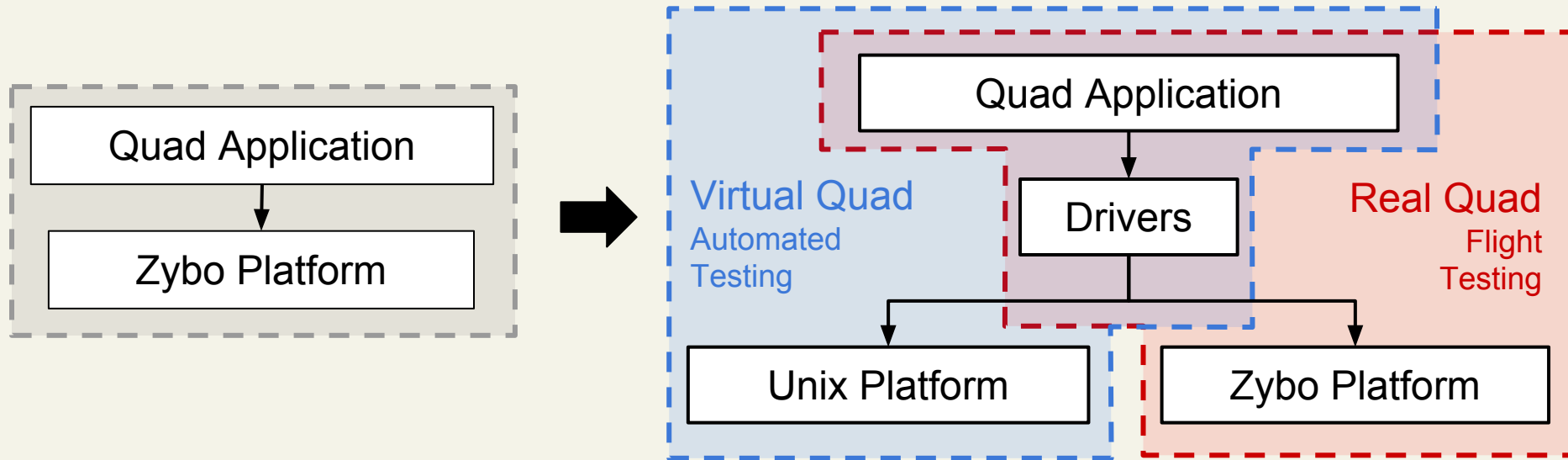
- Previous teams relied on end user tests to verify embedded software
 - But end-to-end tests are expensive in terms of man hours
- Lack of testing flexibility was due to quadcopter software design
 - Tight coupling between the application and Zybo platform
 - Cannot compile for laptops or continuous integration environment



Robustness: Improved Testing Strategy

- Solution

- Re-design software architecture to use interface-like drivers in order to target specific platforms.

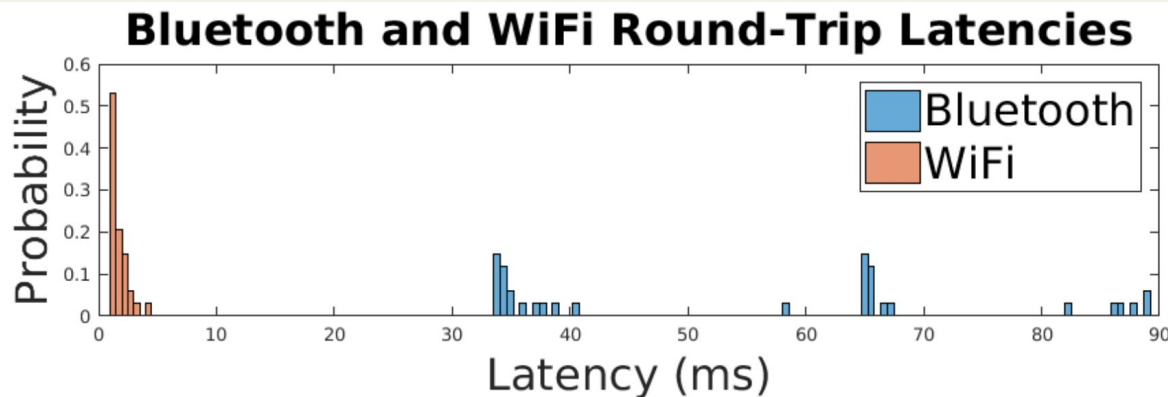
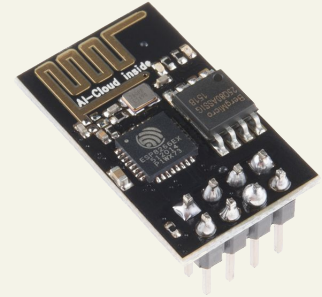


- New Testing Strategy

- **Unit Tests** - Automated on **Continuous Integration**
- **Functional Tests** using the Virtual Quad - Automated on **Continuous Integration**
- **Dedicated Hardware Tests** - Testing each driver manually on the quad
- **End-to-end Tests** - Flying the quad

Robustness: Decreased Latency

- Past issues with autonomy
 - Suspected cause: high latencies
 - Between base station to quadcopter
 - Using Bluetooth
 - 50 milliseconds on average
 - Solution to Decrease latency
 - Communicate via WiFi embedded system
 - Decreased average round-trip latency to 3ms average
 - Increased transmission reliability



Conclusions



Thank You

Questions?

- Team Members
 - Eric Middleton (CprE)
 - Brendan Bartels (EE)
 - Kris Burney (CprE)
 - Andy Snawerdt (EE)
 - Jake Drahos (CprE)
 - Joe Bush (CprE)
 - Tara Mina (EE)
 - David Wehr (CprE)
- Faculty Advisors
 - Dr. Jones
 - Dr. Elia



References

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Budget Plan

Item	Source	Cost
New Groundstation Computer	Provided by Client	\$1400
Frame Kit - DJI Flamewheel F450	Provided by Client	\$190
Optical Flow Sensor	Provided by Client	\$100
Work Lights	Provided by Client	\$70
Tent	Provided by Client	\$100
LiDAR	Provided by Client	\$150
WiFi Module	Provided by Client	\$40
Miscellaneous	Provided by Client	\$50
Total Cost for This Year:	-	\$2100

System Identification

Symbol	Nominal Value	Units	Brief Description
m_q	0.986	kg	Quadrotor mass
m_b	0.204	kg	Battery mass
m	1.19	kg	Quadrotor + battery mass
g	9.81	m/s^2	Acceleration of gravity
J_{xx}	0.0218	kgm^2	Quadrotor + battery moment of inertia around b_x
J_{yy}	0.0277	kgm^2	Quadrotor + battery moment of inertia around b_y
J_{zz}	0.0332	kgm^2	Quadrotor + battery moment of inertia around b_z
J_{req}	4.201210-5	kgm^2	Rotor + motor m.o.i. around motor axis of rotation
K_T	8.155810-6	$kgmrad2$	Rotor thrust constant
K_d	1.747310-7	$kgm2rad2$	Rotor drag constant

System Identification (cont.)

Symbol	Nominal Value	Units	Brief Description
$ r_{hx} $	0.016	m	x-axis distance from center of mass to a rotor hub
$ r_{hy} $	0.016	m	y-axis distance from center of mass to a rotor hub
$ r_{hz} $	0.003	m	z-axis distance from center of mass to a rotor hub
R_m	0.2308	Ω	Motor resistance
K_Q	96.3422	ANm	Motor torque constant
K_V	96.3422	radVs	Motor back-emf constant
i_f	0.511	A	Motor internal friction current
P	0.47	(none)	ESC turn-on duty cycle command
P	0.40	(none)	Minimum Zybo output duty cycle command
P_T	0.80	(none)	Maximum Zybo output duty cycle command
$ r_{hx} $	0.016	m	x-axis distance from center of mass to a rotor hub
$ r_{hy} $	0.016	m	y-axis distance from center of mass to a rotor hub

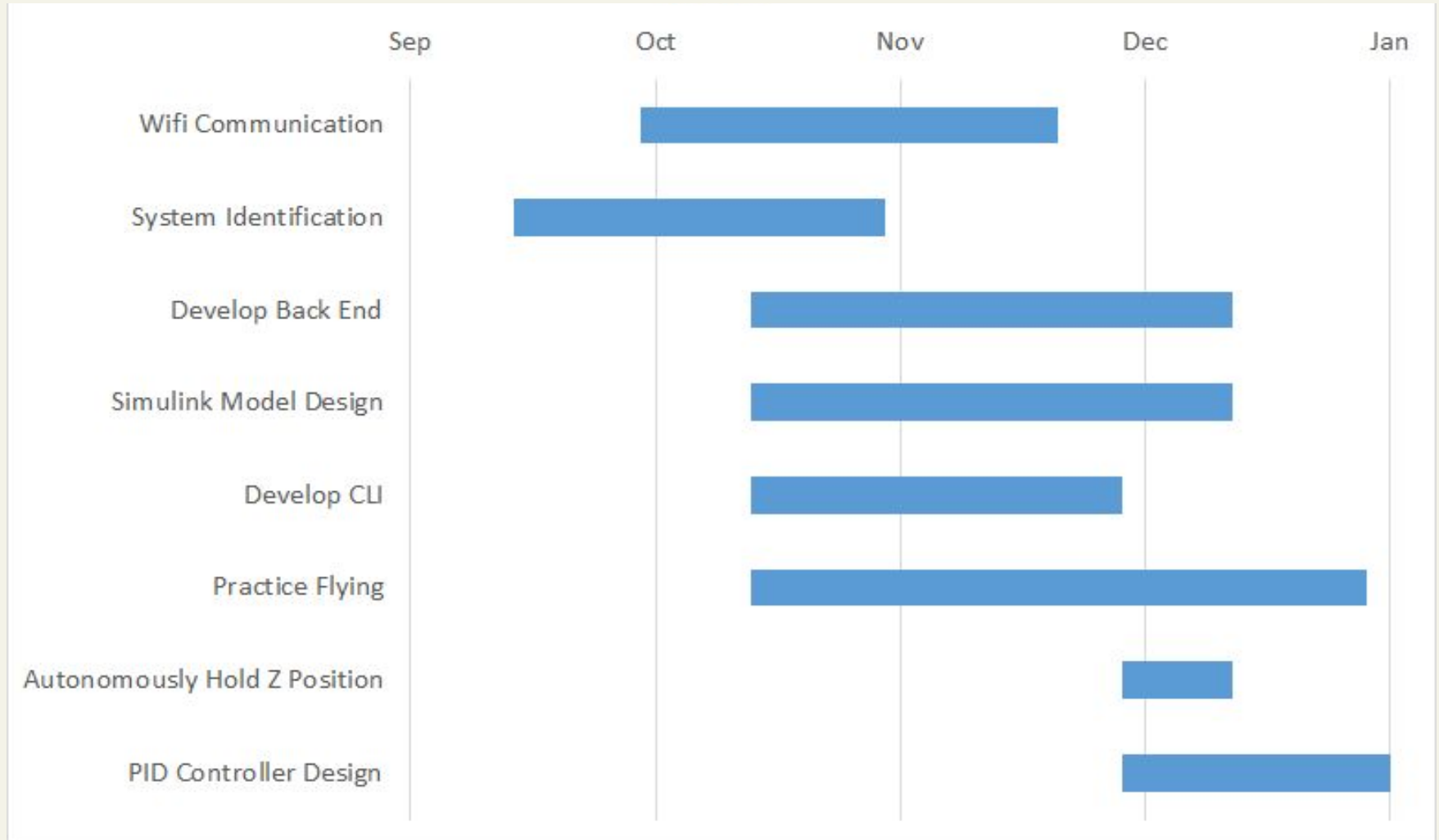
General Safety Practices

- Tether in Flight
- Awareness of Surroundings
 - Respectful of others in lab
 - Observant of obstacles
- Charge batteries in LiPo-safe charging sacks
- Practice Flying Small Quadcopters

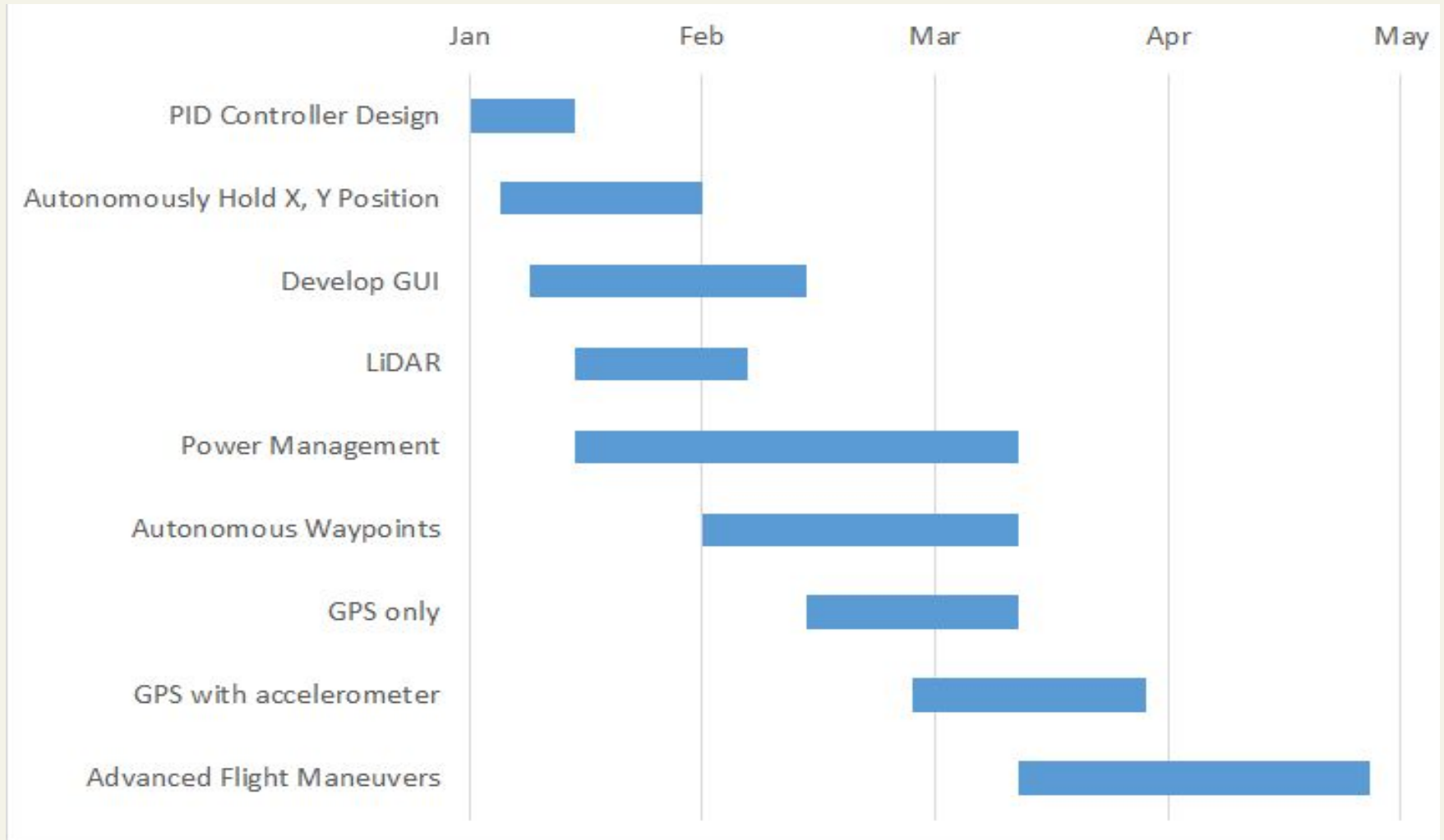
Stages of Testing Software Changes

- Stage 1: Test without Motor Power
 - Can verify that communication & lights work as expected
- Stage 2: Test without Propellers
 - Able to verify that motor velocities are as roughly as expected
- Stage 3: Test with Short Tether
 - Can verify that quadcopter tries to stabilize, and won't fly away
 - Prevents from flipping
 - Emergency: One person holds down quadcopter, another unplugs battery
- Stage 4: Regular Flight Testing
 - Always tethered when in flight

Overall Progress: Fall Semester Timeline



Our Plans: Spring Semester Timeline



Driver Interface Layer

