

# Microprocessor Controlled Aerial Robotics Team (MicroCART)

Dr. Jones and Dr. Elia

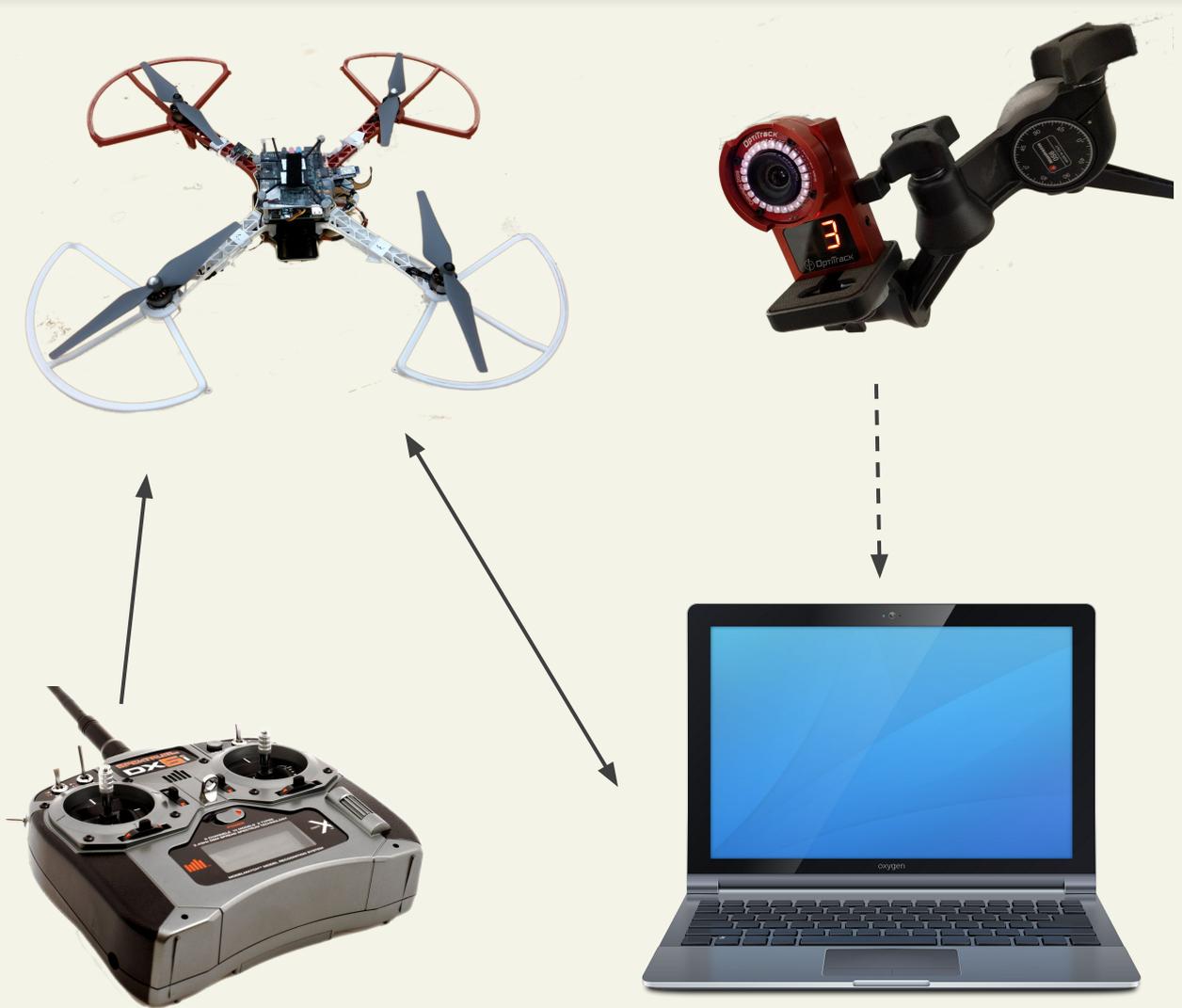
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# Purpose of MicroCART

*To develop an aerial robot as a research platform for controls and embedded systems.*

# System Overview



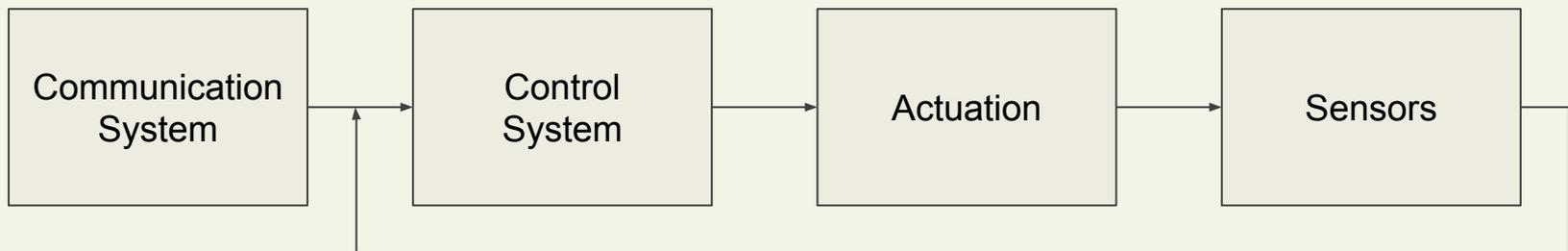
# Goals and Deliverables

- Improved Flight Ability
  - Autonomous Flight
    - Controller designed from mathematical model
    - User-specified waypoints
  - Outdoor flight
- Modular Research Platform Features
  - Customizable controls structure
  - Flexibility in client types (GUI or CLI)
- Increased Robustness of System
  - Continuous Integration and Dedicated Hardware Tests
  - Communication reliability and throughput

# Increased Flight Ability: Mathematical Model

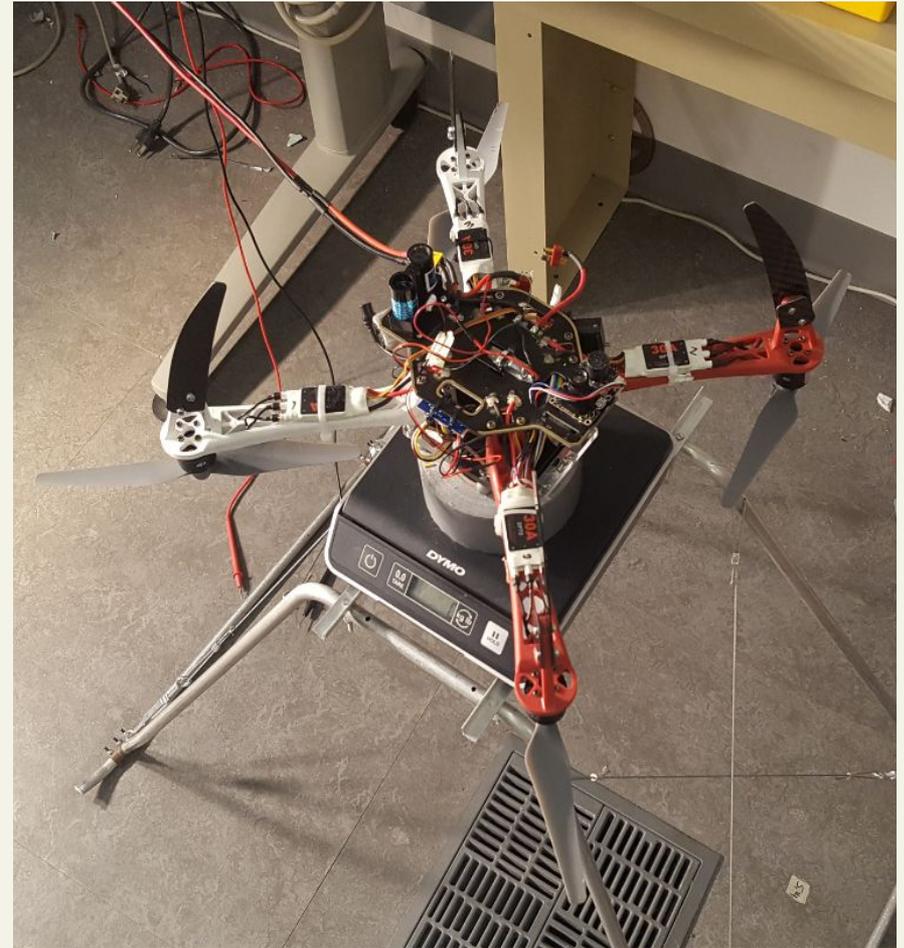
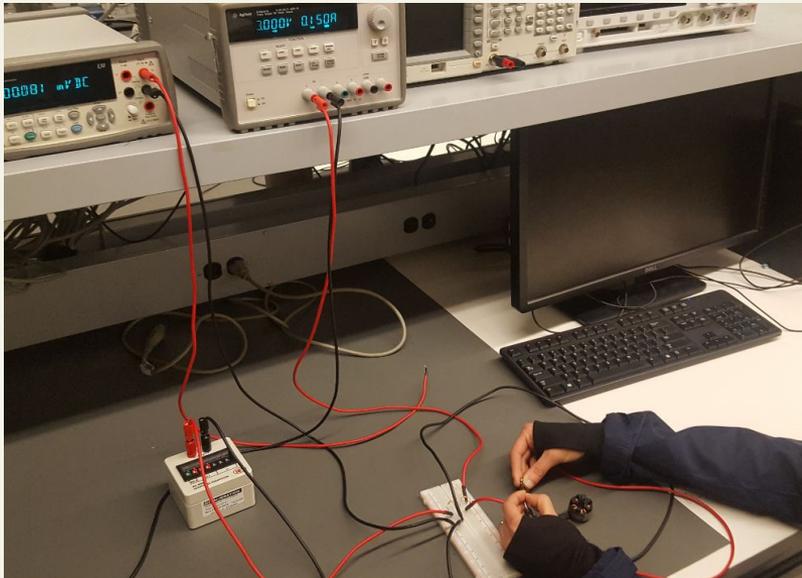
- Previously had no model of current system
  - Advantages:
    - Faster control structure development
      - Allows teams to find stabilizing controllers quickly
      - Different control structures can be simulated before being applied
    - Possibility for more advanced control in the future
      - Model based controllers can be explored

**Quadcopter Model High Level Structure**



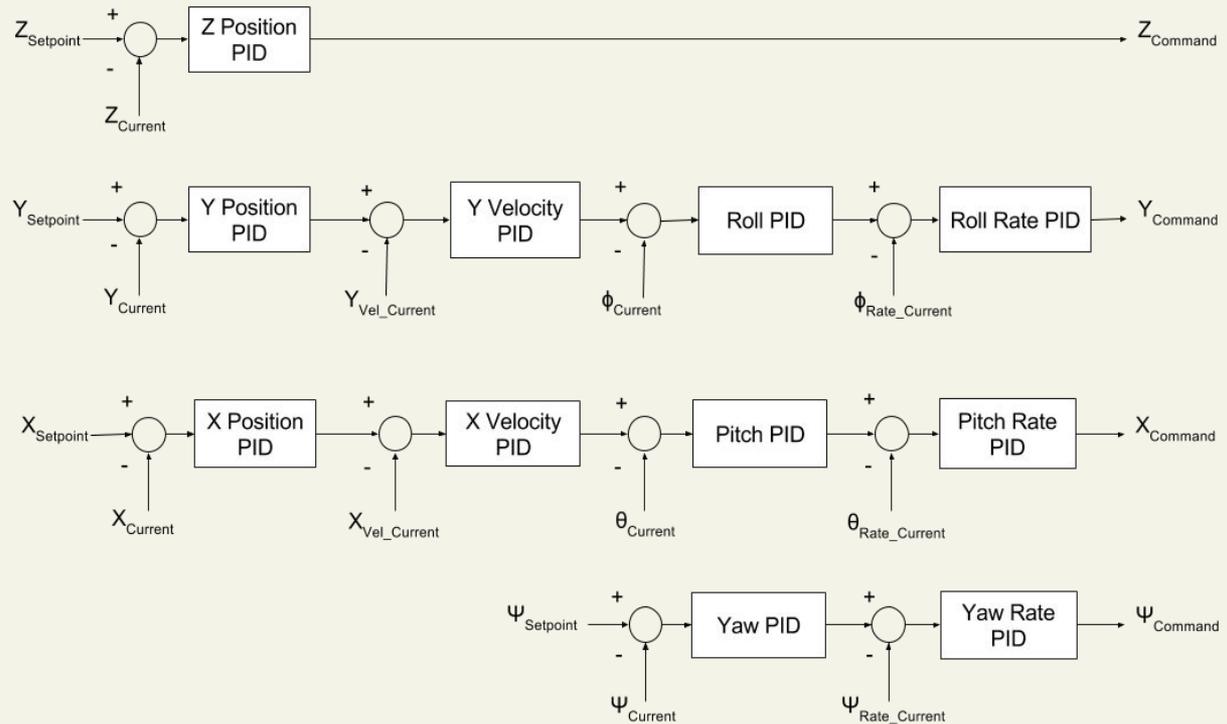
# Increased Flight Ability: Creating the Model

- System Identification
- Parameters Measured:
  - Moments of inertia
  - Thrust and drag constants
  - Sensor noise characteristics
  - Motor resistance



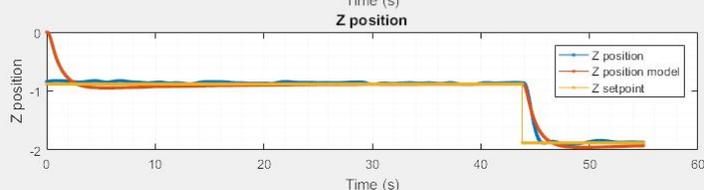
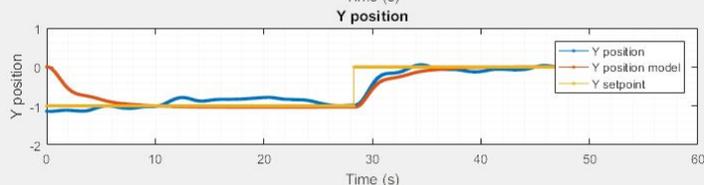
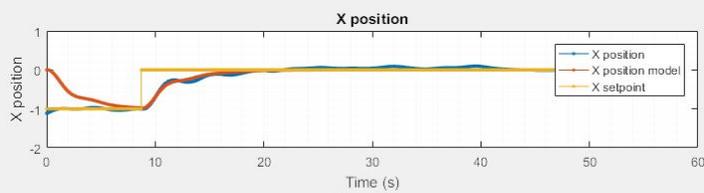
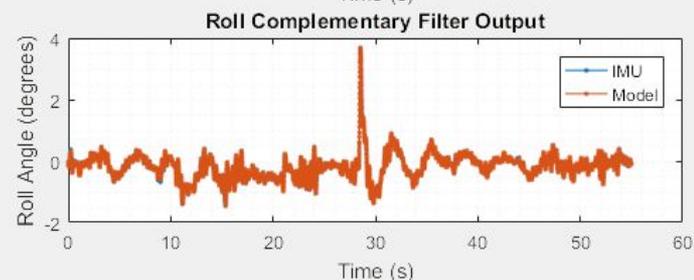
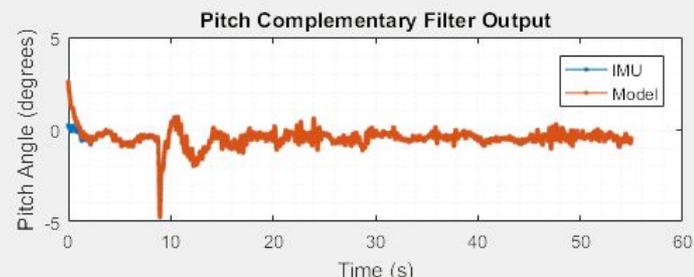
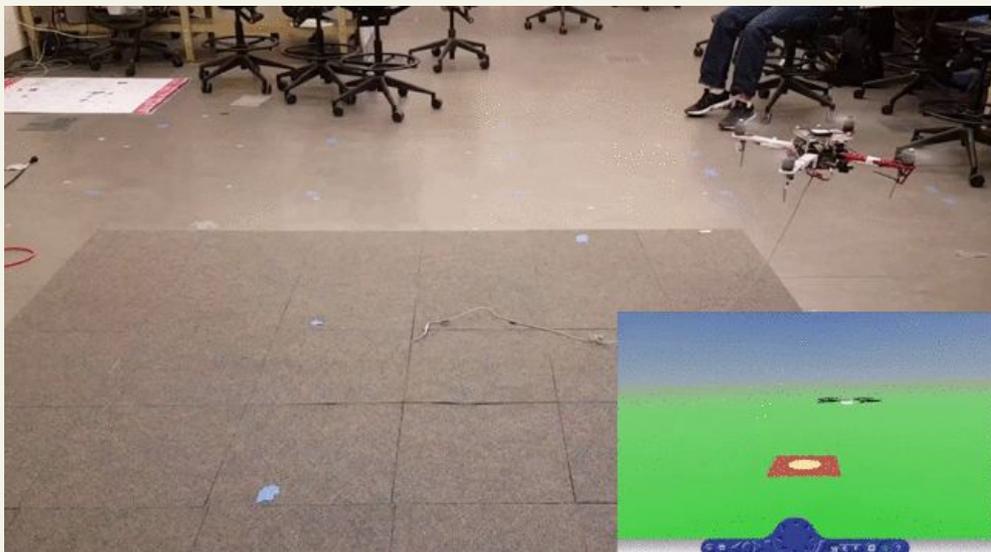
# Increased Flight Ability: Control Structure

- 4 movement options
  - Height
  - Longitudinal
  - Lateral
  - Yaw
- Nested PID Structure
- Position and Velocity Control
- Euler angle and rate control



# Current Model Developments

- Logging Analysis
- Setpoint Testing
  - Current model accurately reflects movement from real quadcopter



# Increased Flight Abilities: Flying Outside

- Flying Outside:
  - LiDAR sensor for distance from ground
    - 1cm resolution
    - Sensor fusion algorithm combines LiDAR and accelerometer data
  - Optical Flow sensor
    - Takes high-speed images of the ground and computes pixel flow
    - Quad computes ground velocities and integrates to estimate position



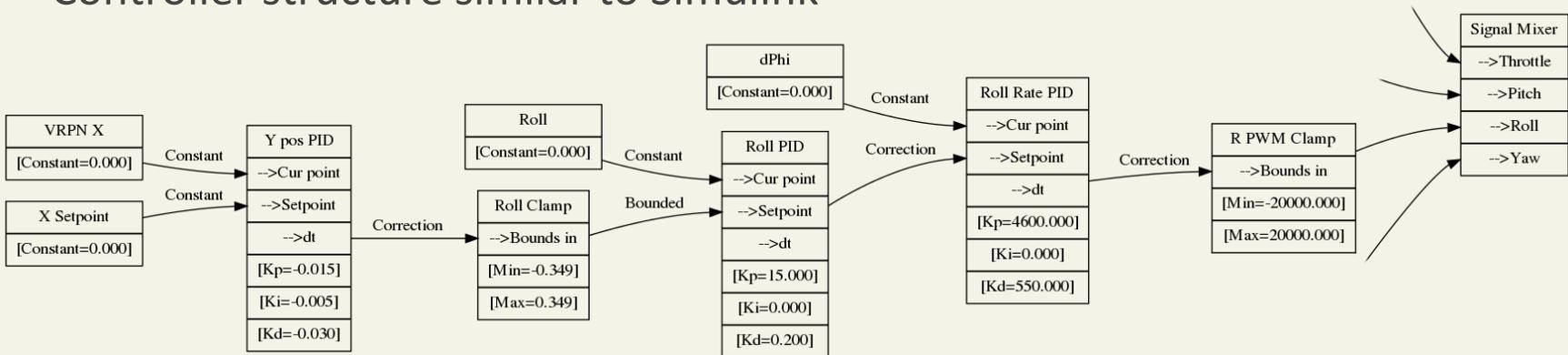
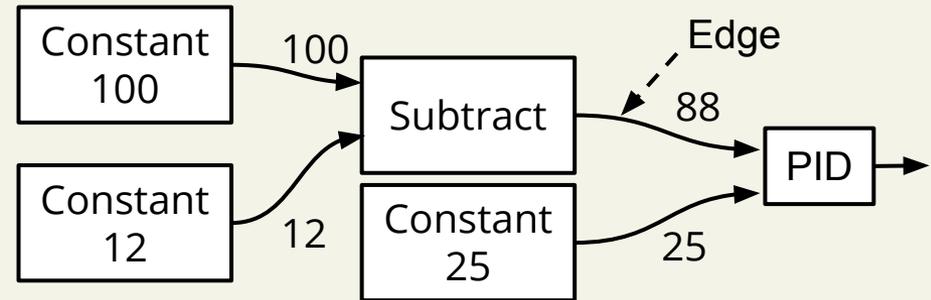
LiDAR Sensor



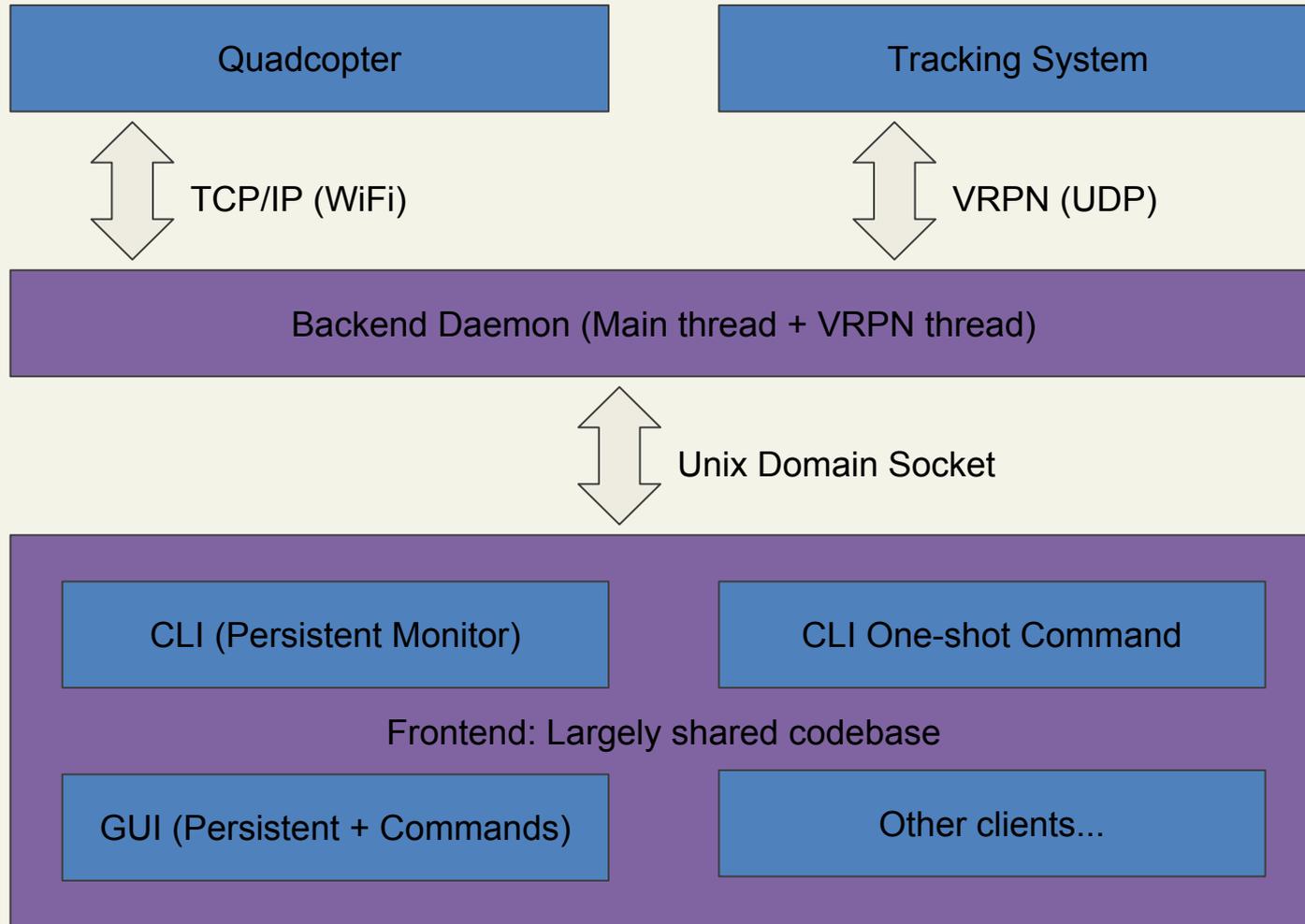
Optical Flow Sensor

# Modular System: Customizable Control Structure

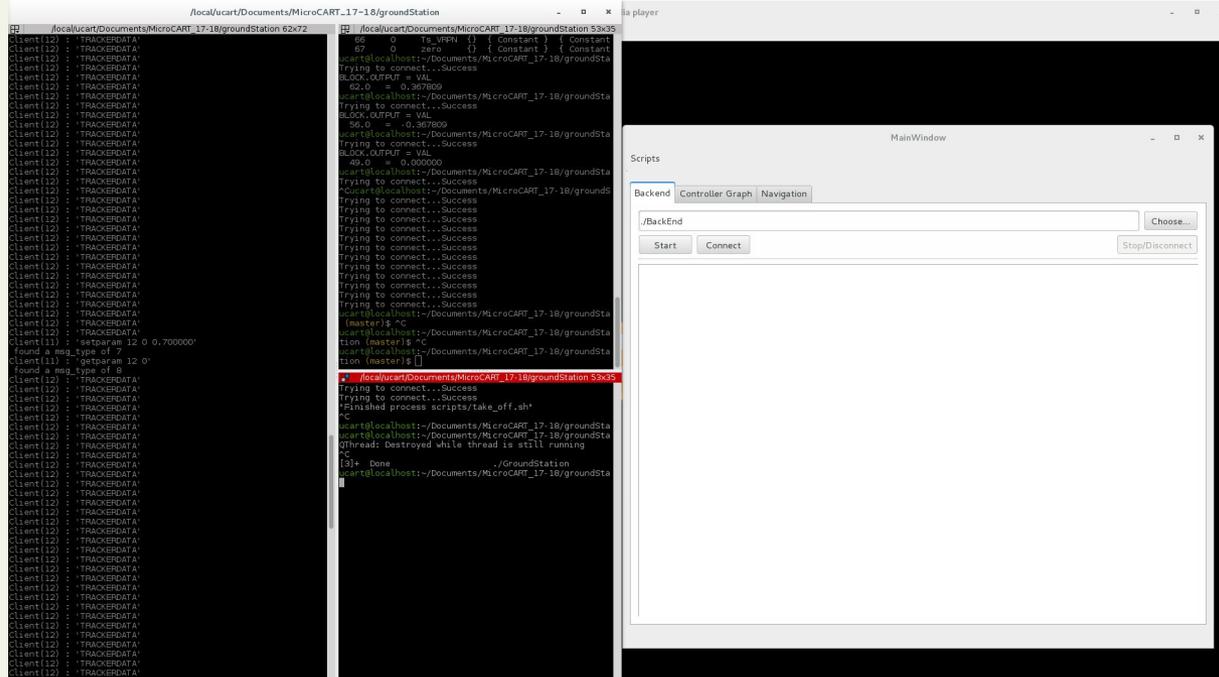
- Structure controller as a directed graph
  - Nodes are discrete functions
  - Calculated values are passed along edges to inputs of other nodes
- Benefits
  - Blocks can be developed and tested independently of the quadcopter system
  - Allows changing controller at runtime
  - Controller structure similar to Simulink



# Modular System: Ground Station Architecture

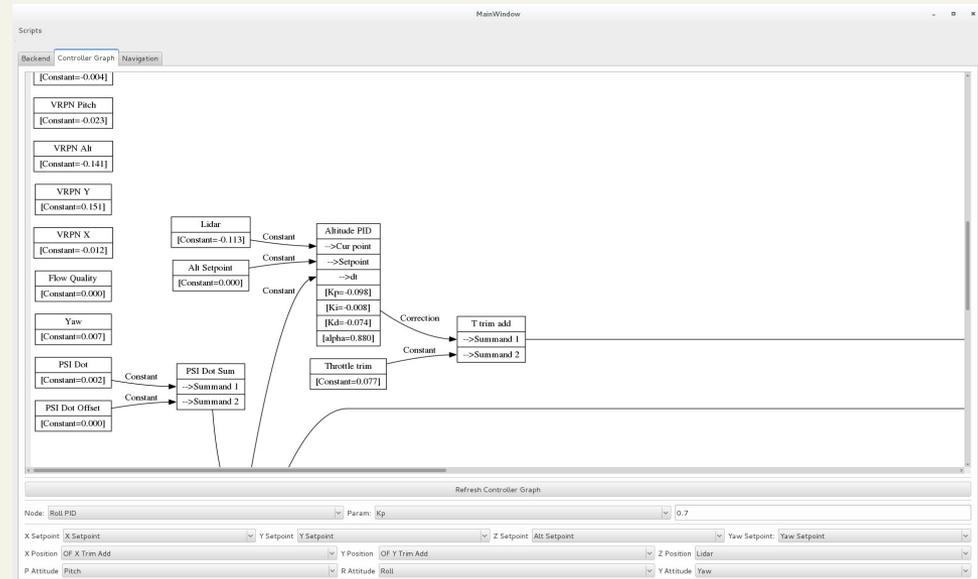
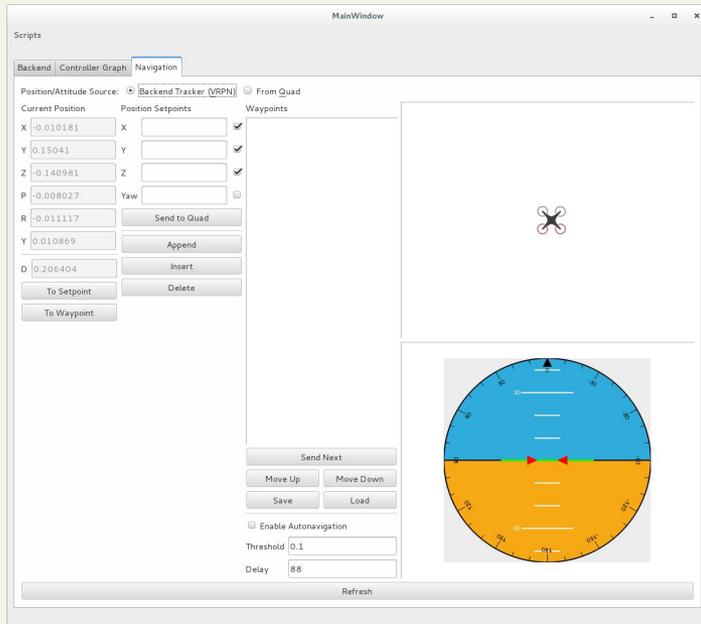
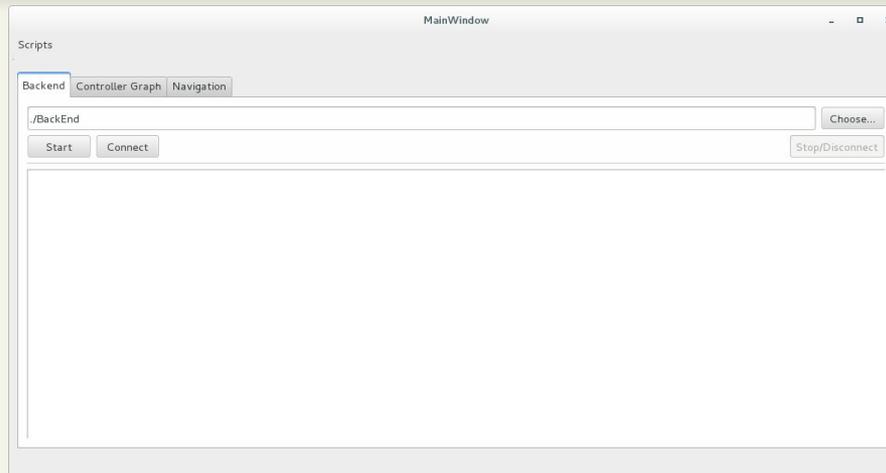


# Ground Station Modular Structure

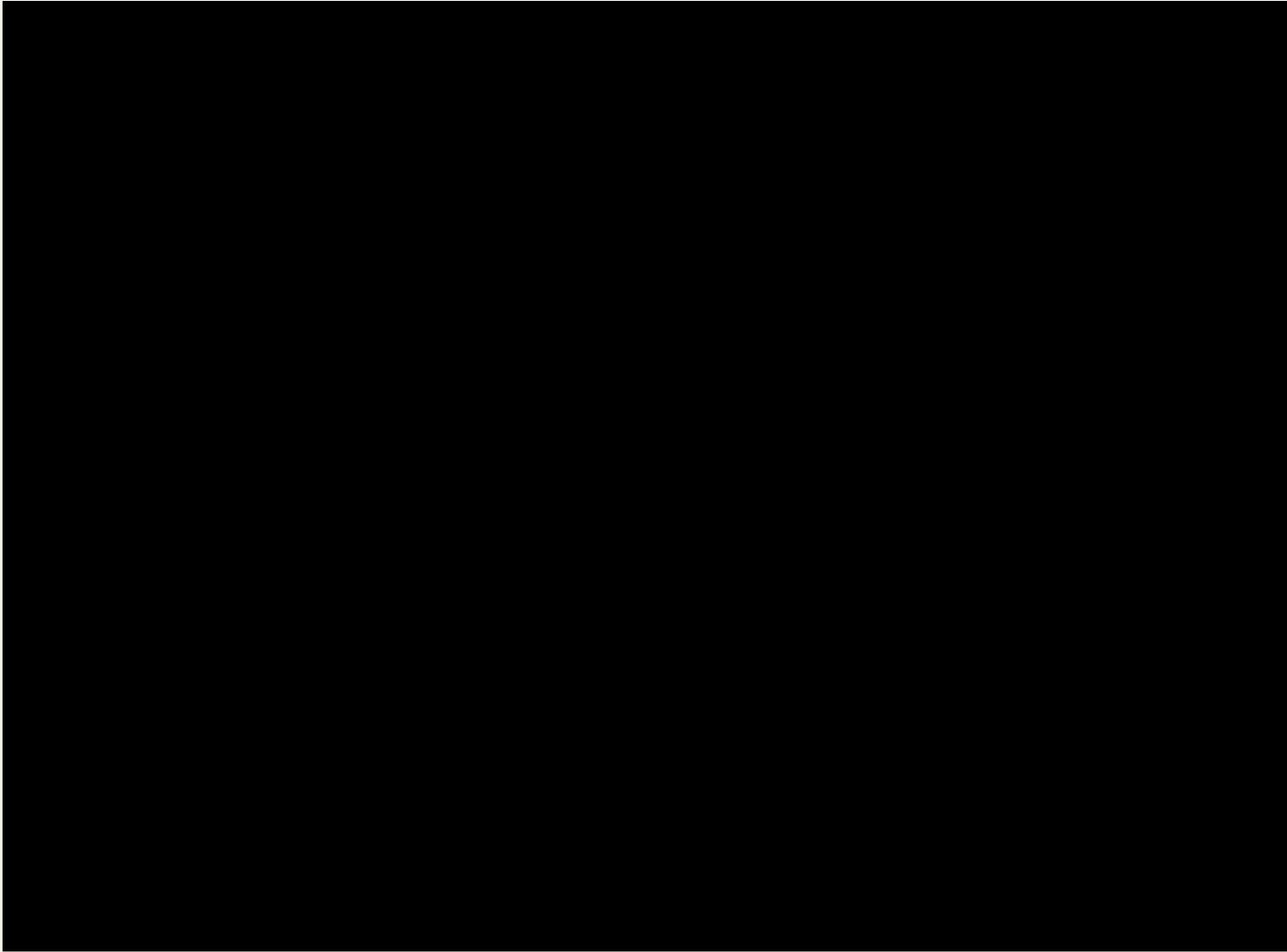


- Backend Daemon
  - Manages quad connection, tracking system
  - Services requests from frontend
- Decoupled Command Line Interface (CLI)
  - getoutput, getparam, getsource
  - setparam, setsource, getnodes
- Intuitive Graphical User Interface (GUI)
  - Same features as CLI
  - More information at-a-glance

# Ground Station GUI

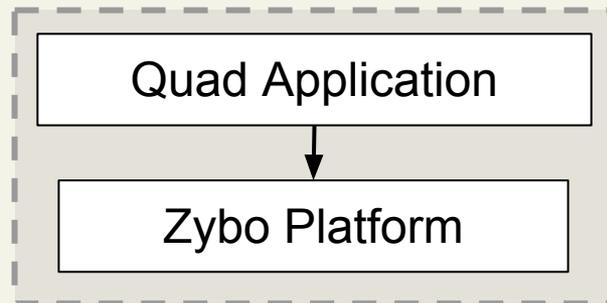


# Ground Station GUI



# Robustness: Improved Testing Strategy

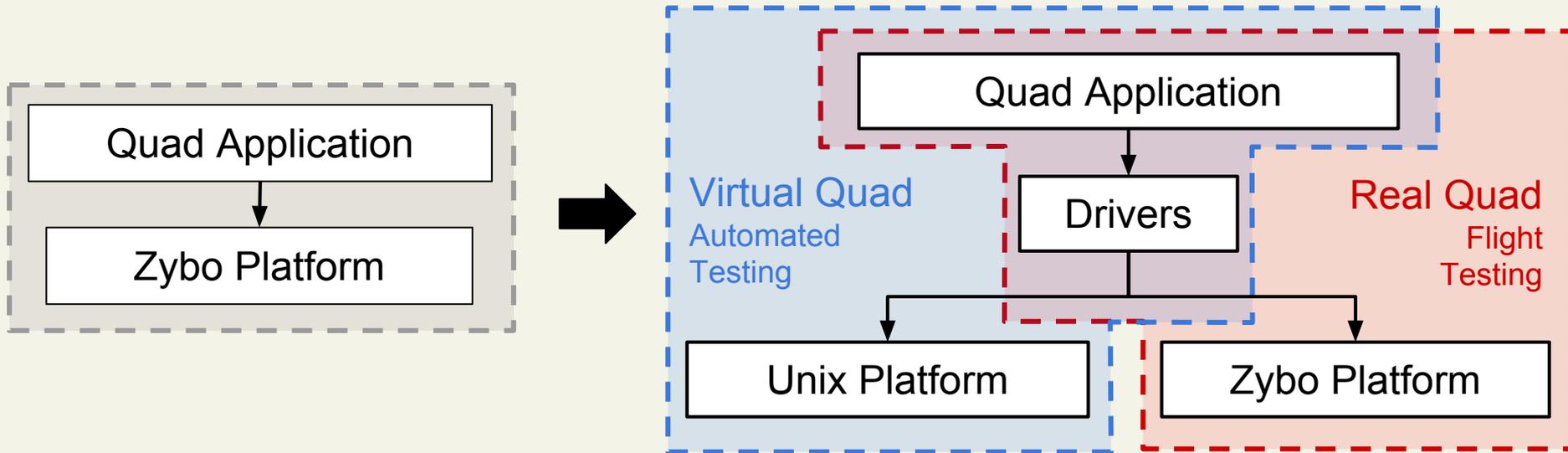
- Problem
  - Previous teams relied on end user tests to verify embedded software
    - But end-to-end tests are expensive in terms of man hours
  - Lack of testing flexibility was due to quadcopter software design
    - Tight coupling between the application and Zybo platform
    - Cannot compile for laptops or continuous integration environment



# Robustness: Improved Testing Strategy

- Solution

- Re-design software architecture to use interface-like drivers in order to target specific platforms.

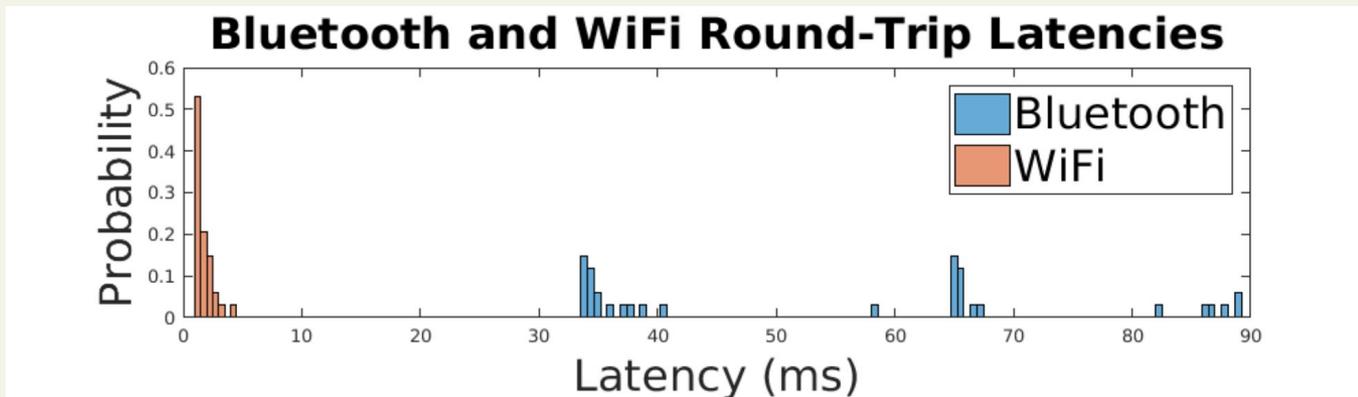
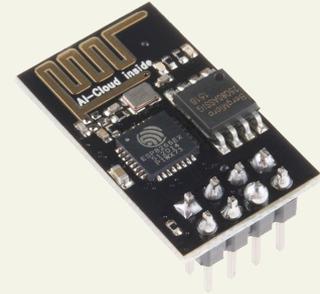


- New Testing Strategy

- **Unit Tests** - Automated on **Continuous Integration**
- **Functional Tests** using the Virtual Quad - Automated on **Continuous Integration**
- **Dedicated Hardware Tests** - Testing each driver manually on the quad
- **End-to-end Tests** - Flying the quad

# Robustness: Decreased Latency

- Past issues with autonomy
  - Suspected cause: high latencies
    - Between base station to quadcopter
    - Using Bluetooth
      - 50 milliseconds on average
  - Solution to Decrease latency
    - Communicate via WiFi embedded system
    - Decreased average round-trip latency to 3ms average
    - Increased transmission reliability



# Conclusions



# Thank You

## Questions?

- Team Members
  - Eric Middleton (CprE)
  - Brendan Bartels (EE)
  - Kris Burney (CprE)
  - Andy Snawerdt (EE)
  - Jake Drahos (CprE)
  - Joe Bush (CprE)
  - Tara Mina (EE)
  - David Wehr (CprE)
- Faculty Advisors
  - Dr. Jones
  - Dr. Elia



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# Budget Plan

Item	Source	Cost
New Groundstation Computer	Provided by Client	\$1400
Frame Kit - DJI Flamewheel F450	Provided by Client	\$190
Optical Flow Sensor	Provided by Client	\$100
Work Lights	Provided by Client	\$70
Tent	Provided by Client	\$100
LiDAR	Provided by Client	\$150
WiFi Module	Provided by Client	\$40
Miscellaneous	Provided by Client	\$50
<b>Total Cost for This Year:</b>	-	<b>\$2100</b>

# System Identification

Symbol	Nominal Value	Units	Brief Description
$m_q$	0.986	kg	Quadrotor mass
$m_b$	0.204	kg	Battery mass
$m$	1.19	kg	Quadrotor + battery mass
$g$	9.81	$m/s^2$	Acceleration of gravity
$J_{xx}$	0.0218	$kgm^2$	Quadrotor + battery moment of inertia around $b_x$
$J_{yy}$	0.0277	$kgm^2$	Quadrotor + battery moment of inertia around $b_y$
$J_{zz}$	0.0332	$kgm^2$	Quadrotor + battery moment of inertia around $b_z$
$J_{req}$	4.201210-5	$kgm^2$	Rotor + motor m.o.i. around motor axis of rotation
$K_T$	8.155810-6	$kgmrad2$	Rotor thrust constant
$K_d$	1.747310-7	$kgm2rad2$	Rotor drag constant

# System Identification (cont.)

Symbol	Nominal Value	Units	Brief Description
$ r_{hx} $	0.016	$m$	x-axis distance from center of mass to a rotor hub
$ r_{hy} $	0.016	$m$	y-axis distance from center of mass to a rotor hub
$ r_{hz} $	0.003	$m$	z-axis distance from center of mass to a rotor hub
$R_m$	0.2308	$\Omega$	Motor resistance
$K_Q$	96.3422	$ANm$	Motor torque constant
$K_V$	96.3422	radVs	Motor back-emf constant
$i_f$	0.511	$A$	Motor internal friction current
$P$	0.47	(none)	ESC turn-on duty cycle command
$P$	0.40	(none)	Minimum Zybo output duty cycle command
$P_T$	0.80	(none)	Maximum Zybo output duty cycle command
$ r_{hx} $	0.016	$m$	x-axis distance from center of mass to a rotor hub
$ r_{hy} $	0.016	$m$	y-axis distance from center of mass to a rotor hub

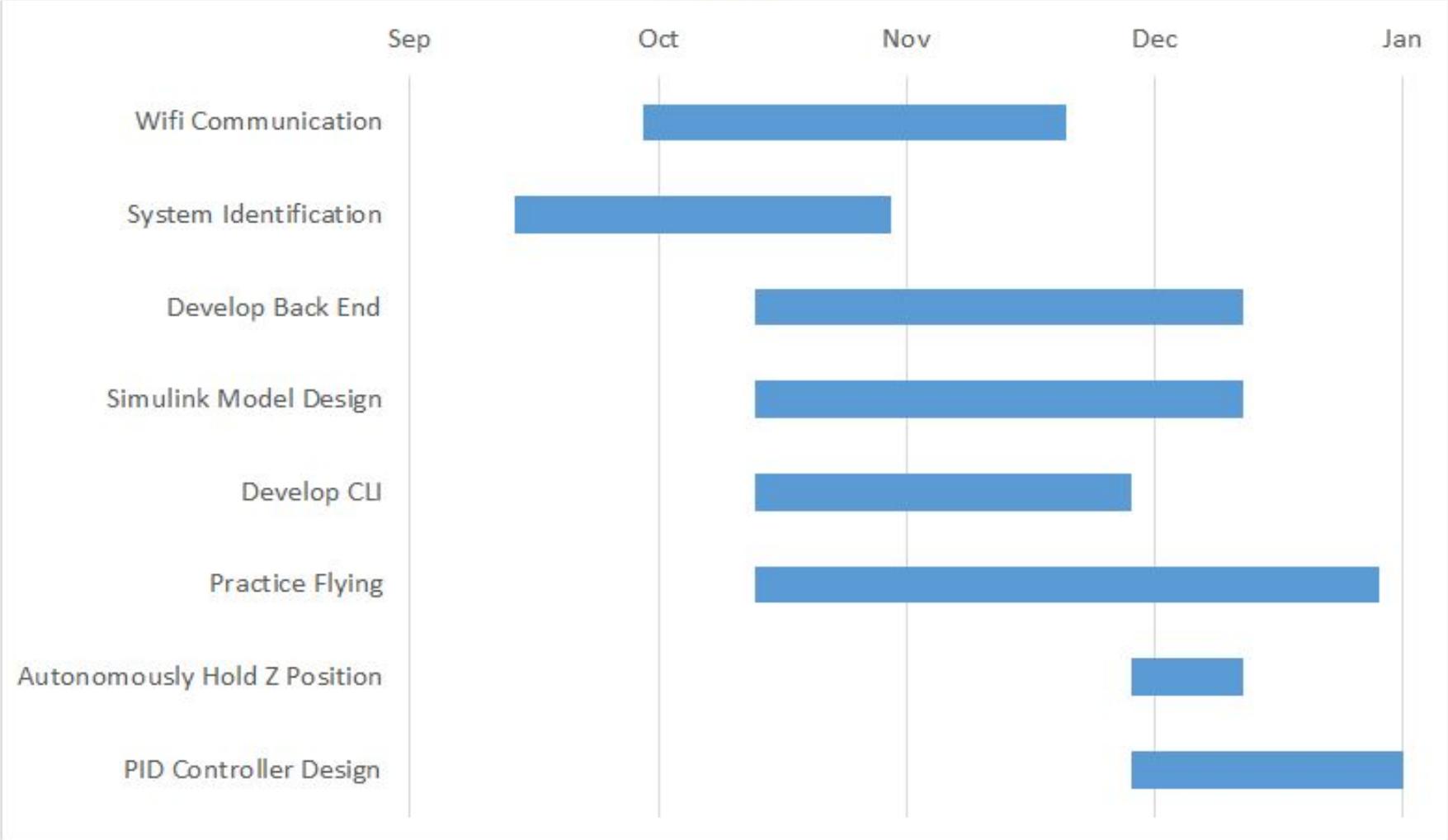
# General Safety Practices

- Tether in Flight
- Awareness of Surroundings
  - Respectful of others in lab
  - Observant of obstacles
- Charge batteries in LiPo-safe charging sacks
- Practice Flying Small Quadcopters

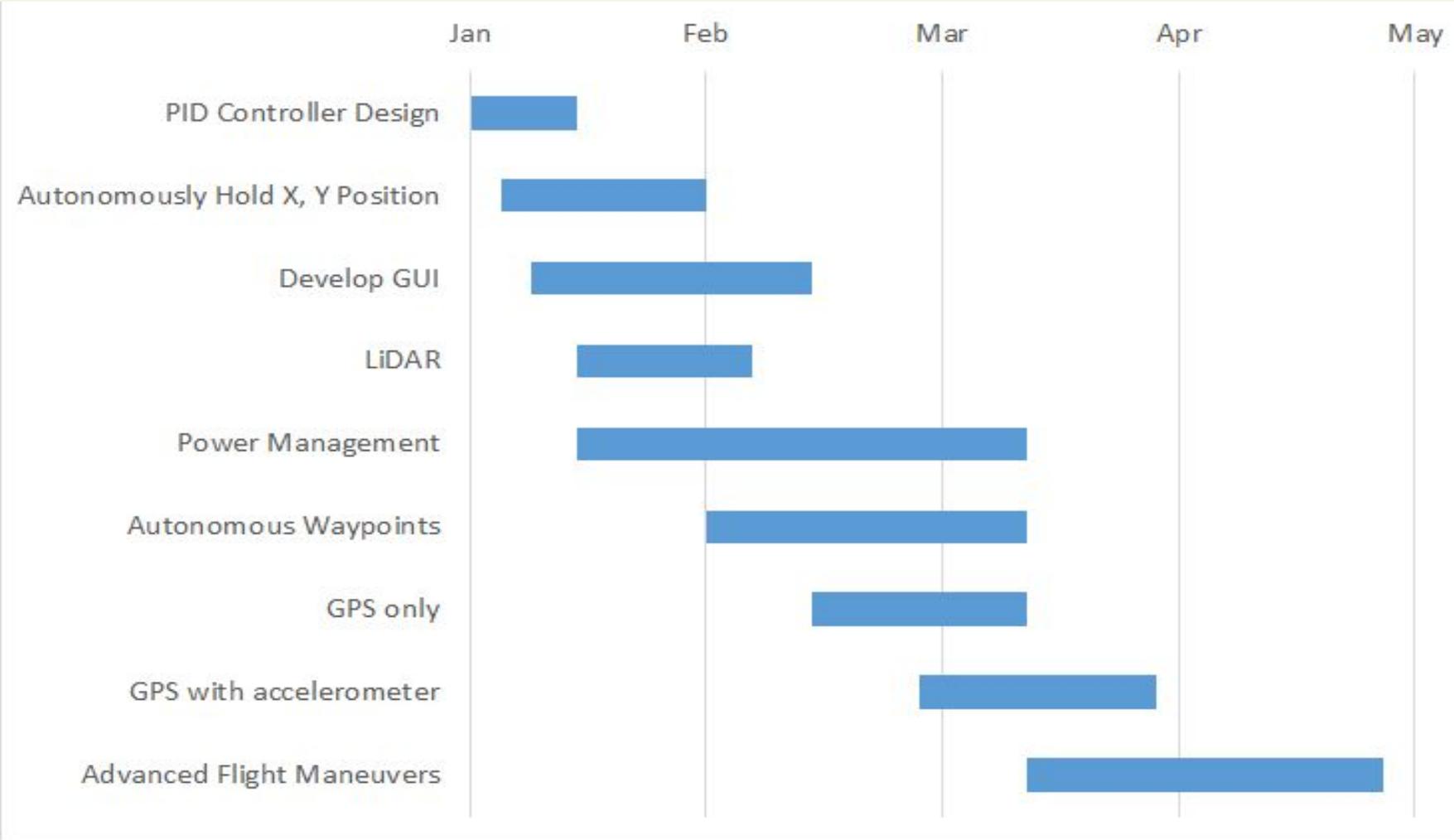
# Stages of Testing Software Changes

- Stage 1: Test without Motor Power
  - Can verify that communication & lights work as expected
- Stage 2: Test without Propellers
  - Able to verify that motor velocities are as roughly as expected
- Stage 3: Test with Short Tether
  - Can verify that quadcopter tries to stabilize, and won't fly away
  - Prevents from flipping
  - Emergency: One person holds down quadcopter, another unplugs battery
- Stage 4: Regular Flight Testing
  - Always tethered when in flight

# Overall Progress: Fall Semester Timeline



# Our Plans: Spring Semester Timeline



# Driver Interface Layer

